Report of the COMSTAC Technology & Innovation Working Group

COMMERCIAL SPACECRAFT MISSION MODEL UPDATE

25 July 1996

Paul Fuller, Chairman Technology & Innovation Working Group Approved for public released
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Commercial Space Transportation Advisory Committee (COMSTAC)
Office Of Commercial Space Transportation

Federal Aviation Administration U.S. Department of Transportation

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Introduction

The U.S. Department of Transportation's Office of Commercial Space Transportation (DoT/OCST) of the Federal Aviation Administration (FAA) endeavors to foster a healthy commercial space launch capability in the United States. An important element of these efforts is to establish the commercial space industry's view of future space launch requirements. Since 1993, the OCST has requested that its industry advisory group, the Commercial Space Transportation Advisory Committee (COMSTAC), prepare and maintain a commercial spacecraft launch demand mission model.

This report presents the 1996 update of the worldwide commercial Geosynchronous Transfer Orbit (GTO) satellite mission model for the period 1996 through 2010. It is based on market forecasts obtained in early 1996 from major spacecraft manufacturers, satellite operators and launch service providers. There are two key points regarding the mission model forecast. First, the mission model is limited to "addressable" payloads, those which are open to internationally competitive launch service procurement. Captive payloads to national flag carriers are excluded from the mission model. Secondly, the number of launches per year resulting from this spacecraft mission model will be a subset of this data due to the potential for multiple manifesting on one or more launch vehicles. The Low Earth Orbit (LEO) and Medium Earth Orbit (MEO) market forecasts have been developed by the DoT/OCST separately from this report (reference 1).

1996 Mission Model Update Methodology

Through a process similar to that in 1995, the Technology and Innovation Working Group solicited input from industry via a letter sent over the signature of the Associate Administrator for Commercial Space Transportation (reference 2). The letter requested market projection data representing the best forecast of the number of addressable commercial GTO payloads per year in the period 1996 - 2010. "Addressable" payloads were defined as those that are open to competitive launch service procurements by U.S., European and other foreign launchers. Excluded were payloads predetermined to be manifested on national flag launch service providers, including government owned payloads, such as DoD and NASA missions, or similar European, Chinese or other nationally captive spacecraft. A table was provided for the respondents to complete which segregated the payloads into categories of "Medium", "Intermediate" and "Heavy" based on separated mass inserted into GTO. In order to accommodate the new launcher initiatives of Delta III and Atlas IIAR, which are considered as Intermediate class given their initial performance characteristics, this class was extended by 1,000 lbs to 9,000 lbs as indicated in the table below. In previous years "Intermediate" was defined as 4,000 lbs to 8,000 lbs.

| Launch Capability (lbs to GTO) | Classification |
|--------------------------------|----------------|
| 2000-4000 | Medium |
| 4000-9000 | Intermediate |
| >9000 | Heavy |

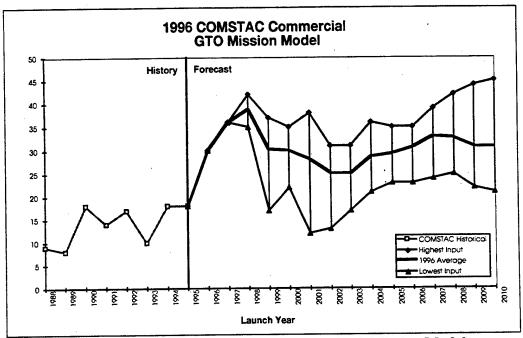


Figure 1.0 1996 COMSTAC Commercial GTO Mission Model

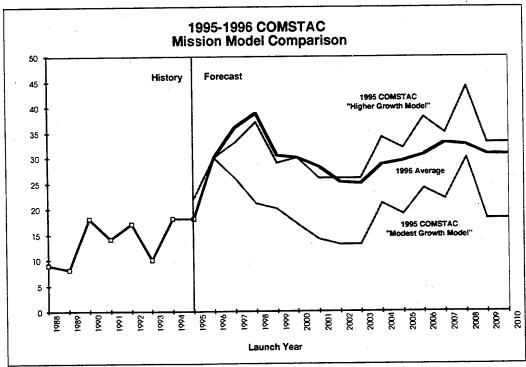


Figure 2.0 1996-1996 Mission Model Comparison

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In 1996, commercial spacecraft launch demand data provided by the following organizations were used in the development of this report:

Boeing Company Lockheed Martin ILS

COMSAT McDonnell Douglas Aerospace

CTA Motorola

GE Americom Orbital Sciences

Hughes Space and Communications Space Systems Loral

INMARSAT TRW

INTELSAT

Conclusions

The following conclusions are based on the results of this 1996 update of the worldwide commercial GTO mission model:

- The 1996 COMSTAC Commercial Mission Model (Figure 1.0) indicates annual demand for launch of commercial GTO payloads will likely be approximately 31 per year in the period 1996 2010. While the high-low dispersions reflect uncertainty in the predictability of the market, it was agreed that the aggregate average is representative of general market forces and trends.
- There is industry-wide convergence toward the 1995 Higher Growth model through 2003. The average of the out year estimates is midway between the 1995 Higher and Modest Growth projections (Figure 2.0). Unlike 1995, the 1996 inputs were not grouped into two distinct higher and modest growth positions. The emergence of new Ka broadband systems and growth in GEO mobile systems may have contributed to the growth in the market forecast.
- The satellite and launch vehicle industry is experiencing a period of significant growth throughout the latter part of the decade. A modest decline beginning in 1998 is expected to be followed by a second wave of growth beginning around 2004 as replenishment of the current generation of active satellites begins.
- The mass of commercial payloads is likely to stay the same or grow, and the mass distribution can best be summarized by two cases:
 - The Stable Mass Growth model predicts that mass trends may have peaked or could stabilize over the next few years.
 - The Continued Mass Growth model predicts steady and continued payload mass growth, generally in line with historical precedent.

These trends are shown in Figure 3.1 and summarized below. In both cases the number of Medium payloads is approximately 11% of the market. The "ILV or HLV" portion of the graph reflects the difference between the two models.

| Stable l Growth | Mass Model | Continu Growth | ied Mass Model |
|--------------------|-------------------|-------------------|-------------------|
| MLV | 11% | MLV | 11% |
| ILV | 70% (23% + 47%) | ILV | 47% |
| HLV | 19% | HLV | 42%(19% + 23%) |
| | OT VESTEE 3247 79 | | |

Average Payload Mass Distribution 1996-2010 (Stable and Continued Mass Growth Models)

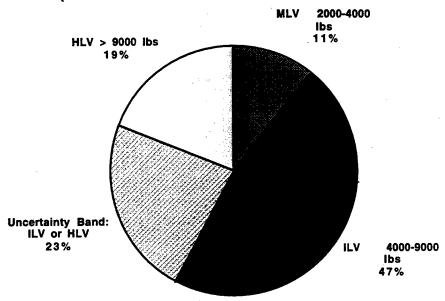


Figure 3.1 Average Payload Mass Distribution 1996-2010, comparing stable and continued payload mass growth models.

Trends in Annual Payload Mass Distribution 1996-2010 (Stable and Continued Mass Growth Models)

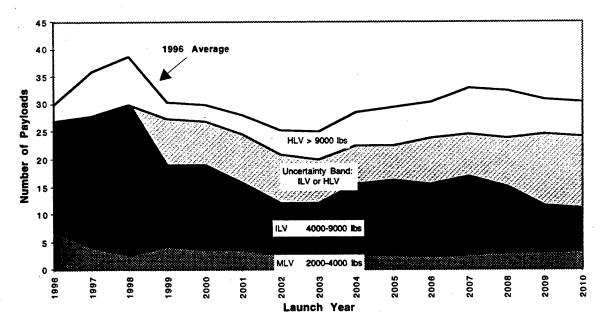


Figure 3.2 Trends in Annual Payload Mass Distribution 1996-2010, comparing stable and continued payload mass growth models.

• In the case of the Continued Mass Growth Model, a significant number of payloads are forecasted to exceed the capability of current U.S. launchers.

Several factors may influence the evolution of payload mass over time (Figure 3.2). These include the introduction of several new heavy-lift launch vehicles, the increased cost effectiveness of larger spacecraft (on a dollars per transponder basis), increasing spacecraft power requirements, and increased orbital congestion. The impact of these factors, along with the potential use of electric propulsion for orbit-raising, may determine whether the market evolves consistently with the Stable Mass Growth case or the Continued Mass Growth case.

Recommendations

The following are the recommendations based on the implications of this 1996 update of the worldwide commercial mission model:

- U.S. launcher programs and initiatives should include a >9,000 lb. capability to maximize commercial market viability.
- The 1996 COMSTAC Mission Model forecast should be provided to appropriate U.S. government agencies for their use.

Reference Tables 1.0, 2.1, 2.2 and 2.3 for complete tabular data.

Please refer to Appendix 1 for the detailed 1996 Discussion and Results, Appendix 2 for the 1996 through 1998 Near-Term Mission Model, and Appendix 3 for the 1988-1995 Payload Launch History.

References

- 1. Department of Transportation letter, "LEO Market Study", dated 2/02/96, F. Weaver
- 2. Depart of Transportation Letter, date 2/20/96, F. Weaver

TABLE 1.0* 1996 COMSTAC COMMERCIAL GTO MISSION MODEL SUMMARY

| CONSTAC 1996 Summary | 26 96 | 26 | 86 | 66 | 00 | 01 | 02 | 03 | 04 | 05 | 90 | 02 | 80 | 60 | 9 | Total 1996- 2010 | Avg 1996- 2010 |
|----------------------------|-------|----|----|----|----|----|----|----|----|----|----|----|----|----|----|------------------------|----------------------|
| Highest Input | 30 36 | 36 | 42 | 37 | 35 | 38 | 31 | 31 | 36 | 35 | 35 | 39 | 45 | 44 | 45 | | |
| Average Rate | 30 | 36 | 40 | 3 | 30 | 53 | 25 | 25 | 29 | 30 | સ | 33 | 33 | 3 | 31 | 464 | 3 |
| Lowest Input | 30 36 | 36 | 35 | 17 | 22 | 12 | 13 | 17 | 21 | 23 | 23 | 24 | 25 | 22 | 21 | | |

* Notes:

- 1) The "Average Rate" 1996 COMSTAC forecast represents the sum of all forecast launch rates divided by the number of all forecasts provided.
- The "Highest" and "Lowest" inputs reflect the maximum and minimum individual estimates provided for any one year. No working group member's forecast was consistently higher or lower than the "Average" throughout the forecast period. 8
- The highest contractor estimate was 523 addressable payloads to be launched from 1996 through 2010. The lowest was 380 and the average was 464. ත
 - 1996 and 1997 figures reflect the consensus forecast developed by the working group members and is provided in detail in Appendix 2, "1996 COMSTAC GTO Commercial Mission Model." 1998 figures represent the highest, average, and lowest mix of potential of launch opportunities consistent with Appendix 2, 1998. 4

TABLE 2.1** FORECAST TRENDS IN PAYLOAD MASS DISTRIBUTION, "MASS **GROWTH STABILIZATION MODEL"**

| Stable Mass Growth Model | 96 | 96 97 98 | 86 | 66 | 99 00 01 | 5 | 05 | 02 03 04 05 | 04 | 02 | 60 80 20 90 | 07 | 80 | 60 | 10 | Total 1996- 2010 | Avg 1996- 2010 | Avg 1996- 2010 |
|--------------------------------|----|----------|----|----|----------|----|----|-------------|----|-------|-------------|----|----|---------|----|------------------------|----------------------|----------------------|
| MLV 2000- 4000 lbs | 7 | 7 4 | ဇာ | 4 | 4 | 4 | ဗ | က | က | | က | က | က | က | က | 53 | 4 | 11% |
| ILV 4000- 9000 lbs | 20 | 24 28 | 28 | 24 | 24 23 | 54 | 8 | 17 | 20 | 20 | 21 | 22 | 21 | 22 22 | 22 | 323 | 21 | %02 |
| HLV > 9000 lbs | က | ھ | 6 | က | ဇ | 4 | 4 | 5. | 9 | 7 | 7 | ω | 6 | 9 | 9 | 88 | 9 | 19% |
| Total | 30 | 30 36 40 | 40 | 31 | 31 30 | 29 | 25 | 25 | 29 | 30 31 | 31 | 33 | 33 | 31 31 | 31 | 464 | 31 | |

* Notes:

1) MLV = The sum of all forecast launch rates divided by the number of all forecasts provided for this mass category.

2) ILV = Total average launch rate less (MLV + HLV). 3) HLV = The sum of the three least aggressive HLV mass growth forecasts divided by three.

TABLE 2.2** FORECAST TRENDS IN PAYLOAD MASS DISTRIBUTION, "CONTINUED MASS GROWTH MODEL"

| Continued Mass Growth Model | 96 | 86 26 96 | | 66 | 8 | 5 | 25 | 03 | 04 | 05 | 90 | 20 | 80 | 60 | 10 | Total 1996- 2010 | Avg 1996- 2010 | Avg 1996- 2010 |
|-----------------------------------|----|----------|-------------|----|----|-------------------|----|----------|----|----|-------------|----|----|-------|----|------------------------|----------------------|----------------------|
| MLV 2000- 4000 lbs | 7 | 4 | က | 4 | 4 | 4 | က | က | က | က | က | က | က | က | က | 53 | 4 | 11% |
| ILV 4000- 9000 lbs | 20 | 20 24 28 | 28 | 15 | 15 | 12 | 6 | o | 13 | 14 | 13 14 13 14 | 14 | 12 | တ | 8 | 215 | 14 | 47% |
| HLV > 9000 lbs | က | 8 | 6 | 12 | F | 13 13 13 13 15 16 | 13 | 13 | 13 | 13 | 15 | 16 | 18 | 19 | 20 | 196 | 13 | 42% |
| Total | 30 | 36 | 30 36 40 31 | 31 | 30 | 29 | 25 | 25 | 29 | 30 | 31 | 33 | 33 | 33 31 | 31 | 464 | 31 | |

** Notes:

Same as Stable Mass Growth Model above. 1) MLV =

Total average launch rate less (MLV + HLV).

2) ILV = Total average launch rate less (MLV + HLV). 3) HLV = The sum of the two most aggressive HLV mass growth forecasts divided by two.

TABLE 2.3*** FORECAST TRENDS IN PAYLOAD MASS DISTRIBUTION, COMBINED STABLE AND CONTINUED MASS GROWTH MODEL COMPARISON

| Continued Stable & Continued Mass Growth Model | 96 | 97 | 86 | 66 | 00 | 6 | 05 | 03 | 04 | 02 | 90 | 20 | 80 | 60 | 10 | Total 1996- 2010 | Avg 1996- 2010 | Avg 1996- 2010 |
|--|----|----------|----|----|----|----------|----|----|----|----|----|----|----|----|----|------------------------|----------------------|----------------------|
| MLV 2000-4000 lbs | 7 | 4 . | က | 4 | 4 | 4 | င | 3 | ဇ | က | က | ဇာ | 3 | 3 | က | 53 | 4 | 11% |
| ILV 4000-9000 lbs | 20 | 24 | 28 | 15 | 15 | 12 | 6 | 6 | 5 | 14 | 5 | 14 | 12 | 6 | 8 | 215 | 14 | 47% |
| ILV or HLV | 0 | 0 | 0 | တ | 8 | 6 | တ | 8 | | 9 | 8 | 8 | 6 | 13 | 14 | 108 | | 23% |
| HLV > 9000 lbs | ო | 8 | 6 | က | က | 4 | 4 | 2 | 9 | 7 | 7 | 8 | 6 | 9 | 9 | 88 | 9 | 19% |
| Total | 30 | 30 36 40 | 40 | 31 | 30 | 29 | 25 | 25 | 29 | 30 | 31 | 33 | 33 | 31 | 31 | 464 | 31 | |

*** Notes:

1) MLV

2) ILV

Same as MLV from both Stable and Continued Mass Growth Models above.

Same as ILV from Continued Mass Growth Model above.

Continued Mass Growth Model HLV forecast less Stable Mass Growth HLV forecast as defined above. 3) ILV or HLV

4) HLV = Same as HLV from Stable Mass Growth Model above.

Examples of Mass Category Definitions:

- Atlas I, Delta II, Long March 3 and 3A, co-passenger payloads on Ariane 4 or 5 and H2 launch vehicles. 1) MLV
- Atlas II, IIA, IIAS, IIAR, Delta III, dedicated H2, Sea Launch Zenit, Proton D1e and co-passenger payloads on Ariane 4 or 5 launch vehicles. <u>|</u>
- Dedicated Ariane 4 or 5, Long March 3B, Proton D1e, and Sea Launch Zenit. 3) HLV

Report of the COMSTAC Technology & Innovation Working Group

COMMERCIAL SPACECRAFT MISSION MODEL UPDATE

May 1996

Appendix 1 1996 Discussion and Results

Commercial Space Transportation Advisory Committee (COMSTAC)
Office Of Commercial Space Transportation
Federal Aviation Administration
U.S. Department of Transportation

1996 DISCUSSION AND RESULTS

The 1996 mission model is made up of two key elements: (1) Payload Launch Demand Forecast: the number of payloads available for launch per year (not adjusted for possible multiple manifesting) and (2) Mass Distribution Forecast: classification of payloads among the launch vehicle categories as described previously in the report. The findings in each of these areas are presented below.

Payload Launch Demand Forecast

A summary of the forecast of payload launch demand GTO Mission Model for the years 1996 to 2010 is shown Figure 1.0 (refer to the body of the report for all figures). The display represents the aggregate average of all responses plotted against the range of dispersions. The dispersions reflect the highest and lowest data points in each given year. The 1996 forecasts of the numbers of payloads to be launched varied greatly, as the dispersions suggest, however, there was convergence among a significant number of working group members around the average, particularly in the early years of the forecast period. It is important to note that the 1996 data forecasts were not polarized around two viewpoints regarding market growth, as in the 1995 model. Consequently, there was agreement among the participants that the aggregate average can be used as a representative means of portraying the general market forces and trends.

The approach used by the industry to forecast the commercial satellite demand includes evaluating firm contracted missions, current satellite operators planned missions, current operator's replacement missions, current operators growth and growth replacement, as well as "attrition" and some assessment of "unidentified" growth. Attrition is assumed to be 10% of annual launch demand. It includes on-orbit satellite and launch vehicle failure rate, with the demand replaced two years after failure. Unidentified growth includes proprietary, company specific information on future market demand. Differing opinions on the unidentified growth play a key role in the variance in the data toward the outyears of the model.

The plot of the 1996 aggregate average forecast against the 1995 model is depicted in Figure 2.0. The data suggests that since the time of the 1995 report, there is a greater consensus and more confidence in the 1995 Higher Growth model through the early 2000 timeframe. The forecast then remains about midway between the two 1995 cases for the remainder of the period. Events such as the Ka broadband program FCC filings in October 1995 and the emergence of GEO mobile systems may underlie this change since the 1995 report.

The number of payloads actually launched in 1995 (18) was lower than the demand predicted (22) in the 1995 report. A mission by mission assessment indicates that 4 payloads were delayed from 1995 into early 1996, two on Ariane, one on Delta and one on Long March. Typically, these delays are due to a combination of effects including failure-related launcher delays, satellite readiness, and customer preference.

The average rate over the 15 year planning period represents an annual demand of 31 payloads per year (Table 1.0). This is slightly below the 1995 Higher Growth average of 31.8, but significantly above the Modest Growth average of 20.5. The effect of averaging also results in a smoother forecast than that portrayed in 1995, however, the cyclical nature of this market is still noticeable in the data. A dip in the early 2000 timeframe is followed by resurgence of demand in the period 2003 to 2008, typically reflective of replenishment requirements, which is consistent with, but not as marked as in the 1995 forecasts.

The 1996 Arianespace mission model published in March (reference 1) included both Maximum and Nominal forecasts of the period 1996 through 2003. Both are significantly higher than their previous estimates of an average of approximately 15 payloads per year. The 1996 Arianespace Maximum generally tracks closely with the COMSTAC 1996 industry average although their

forecast is slightly lower than COMSTAC in the first two years; the Nominal forecast is generally 5-10 payloads lower than the 1996 COMSTAC average in the years 1999-2003.

Graphs and Calculations (Demand)

The 1996 COMSTAC Mission Model was based on all of the inputs received from the 13 organizations listed in the Methodology. Of those inputs, five inputs were comprehensive and covered the entire addressable commercial GTO market. The remaining inputs were used to verify missions and dates in the five comprehensive mission model forecasts. Accordingly, these five comprehensive mission models provided the basis for all the calculations summarized in this report.

The average launch rate from 1996 through 2010 rate was calculated by adding the five comprehensive working group forecasts together and dividing them by five (Figure 1.0 and Table 1.0). Estimates for 1996 and 1997 reflect the consensus forecast developed by the working group and are provided in detail in Appendix 2, "1996 COMSTAC Commercial GTO Mission Model."

Estimates for 1998 reflect varying estimates of launch rate demand, with a high of 42 and a low of 35, out of the total of 48 potential programs listed in Appendix 2. These differences stem from independent assessments of the likelihood or timing of the unassigned or spacecraft "not ordered" programs annotated in Appendix 2.

The highest and lowest inputs (shown in Figure 1.0 and Table 1.0) represent the single highest or lowest estimated number of payloads to be launched in that year. No working group member's forecast was consistently higher or lower than the "Average" throughout the forecast period. Therefore, the maximum inputs and minimum inputs are not additive. Accordingly, the highest single cumulative estimate across the 1996-2010 forecast period was 523 addressable commercial payloads to be launched. The lowest cumulative estimate was 380 and the average was 464.

Mass Distribution Forecast

Figures 3.1 and 3.2 reflect a significant issue addressed by the working group: How far and how fast will trends in commercial satellite payload mass evolve?

The working group had two answers to the question, with one group predicting that payloads are likely to continue to grow, generally in line with their historical track record, and another group advocating that weight trends may have peaked or could stabilize over the next few years.

It is interesting to note that in either case, in contrast to many U.S. government-funded programs where funding considerations may drive payload mass down, the working group agreed that commercial satellites are likely to stay roughly the same size or even grow.

Some continued growth advocates suggested that a significant, unanticipated market correction was underway and that one-half to two-thirds of addressable commercial payloads could weigh well over 9,000 lbs within the next five to ten years. Other continued growth advocates suggested a more moderate evolution, with about one-third of all new programs weighing over 9,000 lbs by the year 2000. A steady, but gradual evolution could occur thereafter, with roughly half the market weighing over 9,000 lbs by 2006.

Likewise, some members of the "stable growth" group believed that there would be little to no growth, while others advocated that there was likely to be at least some growth, but not necessarily significant. In all cases some payload mass growth is masked by the broadening of the intermediate payload class from 4,000-8,000 lbs to 4,000-9,000 lbs.

Continued growth advocates pointed out, however, that there was more than one reason satellites could continue to grow. Principle among their arguments is the commercial introduction of several new and powerful international launch service systems: the French Ariane 5, the Russian Proton, the Chinese Long March 3B and the Sea Launch program. Each of these systems may not prove successful in its own right as planned. However, experience indicates that over time, the demand for larger satellites follows the supply of larger rockets. Therefore, even if priced at parity with U.S. launch vehicles, these new rockets could usher in a strong new competitive challenge to medium and intermediate U.S.-based launch service providers. This, however, could be offset by the impact of trade agreements and reliability concerns.

Other rationale considered by continued growth advocates includes:

- Affordability: Comparing the product of satellite power, bandwidth and satellite lifetime and dividing by cost, a very large "condominium" satellite can cost significantly less than a smaller spacecraft.
- Power: End user demand for more satellite power is increasing rapidly either for direct broadcast satellites (smaller dishes and better signal quality); for geosynchronous mobile communications satellites (more voice circuits and power hungry digital signal processing); and finally, for next generation broadband multimedia interactive satellites (higher data rates, smaller dishes, and better signal quality).
- Orbital Congestion: The geosynchronous orbital arc is steadily becoming more crowded and fewer transmit frequencies are available for use. This can result in heavier spot beam antennas and frequency reuse (for Ka-band satellites) or demanding orbital separation requirements (and ultimately larger and more capable satellites) for co-coverage C- and Ku-band applications.

Another difference of opinion within the working group relative to growth of payload mass was their forecast of the relative commercial success of electric propulsion for orbit raising. Some group members believed that there had been, and would likely remain limited commercial interest in this technology. Others suggested that some, but perhaps not all commercial customers could stand to benefit from it. The group agreed, however, that while electric propulsion systems are currently in use for satellite station keeping, they have not significantly slowed the growth in payload mass. On the other hand, if electric propulsion for orbit raising proved commercially successful, it could reduce propellant mass significantly.

An implication of the Continued Mass Growth Model is the potential loss of market share by U.S. launch providers, given no current U.S. launch capability greater than 9,000 lbs.

Graphs and Calculations (Payload Mass)

Both the stable and continued payload mass growth models are identical in several respects. For purposes of comparison, both models are based on the same average launch rate estimate from 1996 through 2010 (Figure 1.0 and Table 1.0) discussed earlier in the report. Likewise, both models are divided into three different payload mass categories: Medium, Intermediate and Heavy (Tables 2.1 and 2.2). Finally, both models contain the same estimate of Medium payloads to be launched in a given year, reflecting a general consensus within the group regarding the number of future launches in this weight category. This estimate therefore represents the sum of all five forecasts for this mass category divided by five.

The two payload mass distribution models differ thereafter. For example, the Intermediate and Heavy payload mass estimates for the continued mass growth model are based on the average of the two forecasts that suggest the most change in payload mass. Likewise, the corresponding Intermediate and Heavy payload mass estimates for the stable mass growth model are based on the average of the three remaining forecasts (Tables 2.1 and 2.2).

The "pie" chart (Figure 3.1) and the corresponding stacked "area" chart (Figure 3.2) were calculated as outlined below. The 1996-2010 averages are based on the corresponding sum of estimates covering the entire forecast period. Similarly, the annualized estimates (Figure 3.2) are based on year-by-year totals only. As stated earlier, the calculations are based on the five comprehensive commercial GTO forecasts.

The Uncertainty Band "ILV or HLV" portion of each graph reflects the difference between the two mass growth models, an average of 23%.

| Number | Calculation Method |
|---------------------------------------|--|
| Total number of payloads | All 5 comprehensive forecasts divided by 5 |
| MLV payloads | All 5 MLV forecasts divided by 5 |
| HLV payloads | 3 "Stable Payload Mass Growth" forecasts divided by 3 |
| Uncertainty Band: HLV or ILV payloads | 2 "Continued Payload Mass Growth" forecasts divided by 2 Less (HLV payloads) |
| ILV payloads | Total number of payloads Less (MLV + HLV + "HLV or ILV" payloads) |

Background

COMSTAC prepared the first commercial mission model in April 1993 as part of a report on commercial space launch systems requirements (reference 2). Each year since 1993, COMSTAC has issued an updated model. The process has been continuously refined and industry participation broadened each year to capture the most realistic portrayal of space launch demand possible. Thus, the COMSTAC mission model has been well received by industry, government agencies and international organizations.

1993: The first report was developed by the major launch service providers in the U.S. and covered the period 1992 - 2010. The report projected only modest growth in telecommunications markets based mainly on replenishment of existing satellites, with only limited new satellite applications.

1994: Key U.S. spacecraft manufacturers contributed to the 1994, report which represented an average of inputs by Hughes Space & Communications, Martin Marietta Astro Space and Space Systems Loral. The demand reflected an average of 17 payloads per year over the forecast period of 1994-2010, with some members of the spacecraft manufacturing community believing the mission model to be conservative.

1995: The Technology and Innovation Working Group was formally chartered to prepare an annual Commercial Spacecraft Mission Model Update Report (reference 3, 4). The organizations from which the market demand forecasts were requested was further expanded to include satellite operators, in addition to spacecraft manufacturers and launch service providers. The 1995 data contained sizable variations in projected launch demand with a significant degree of polarization around two differing viewpoints. Therefore, a two case scenario was adopted for the 1995 report. A "Modest Growth" scenario projected an average demand for launch of approximately 20 payloads per year over the period 1995 to 2010. A "Higher Growth" scenario forecasted the demand to be an average of 32 payloads per year. Both models included firm contracted missions, satellite operator's planned missions, growth, replenishment and attrition (replacement for launch or on-orbit failure, assumed to be 10% in total). The "High Growth"

case also included a segment called "unidentified growth" often based on proprietary information from the survey respondents.

In the 1995 model there was general agreement among the participants regarding the distribution of payloads among the different weight classes. In both the Modest and Higher Growth cases approximately 70% of the payloads were forecasted to be in the Intermediate category (4000 - 8000 lb.), with 15% each in the Medium (2000-4000 lb.) and the Heavy (>8,000 lb.) classes.

Appendix 1 References

- 1. "Space Transportation Markets", Arianespace, March 1996
- 2. COMSTAC Report, "Commercial Space Launch Systems Requirements 28 April 1993", Office of Commercial Space Transportation, U.S. Department of Transportation, Washington, D.C.
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Report of the COMSTAC Technology & Innovation Working Group

COMMERCIAL SPACECRAFT MISSION MODEL UPDATE

May 1996

Appendix 2 1996 - 1998 Near-Term Mission Model

Commercial Space Transportation Advisory Committee (COMSTAC)
Office Of Commercial Space Transportation
Federal Aviation Administration
U.S. Department of Transportation

1996 Mission Model - Near Term

Near Term Payload Launch Demand Forecast 1996 through 1998: A summary of the near-term 1996-1998 mission model individually identified by name is presented in Appendix 2. The table is divided into addressable commercial GTO spacecraft and non-commercial spacecraft that will potentially utilize the same commercial launch systems. The non-commercial spacecraft forecast includes payloads captive to national flag carriers. There has recently been an initial breakthrough of U.S. spacecraft manufacturers into this market, and there is speculation that the launch service segment of this market may eventually open to U.S. competition, although doubtful until beyond 2000. In the period through 1997, most launch procurement decisions have been made and the launch vehicle manifests have been established. Over this time period, satellite lead times are striving for 12-18 month delivery cycles, while launch vehicles deliveries remain closer to 24 months. Therefore, pressure continues for launch vehicle manufacturers to compress production and/or cycle times.

Note, however, that even in this near-term period complete unanimity was not reached due to differences in opinions on outcomes of expected demand including effects of double booking, program delays, etc. Therefore, the ground rules that were adopted to arrive at the forecast presented are stated below:

- Published manifests of the launch service providers were used unless a failure event of other recognizable event has caused a delay.
- Where manifests do not exist, or where an event which caused a delay has occurred, the subgroup relied on the data source within the subgroup that most likely had the superior knowledge. For example, the McDonnell Douglas representative could modify the published manifest data for the Delta II, or a spacecraft manufacturer with knowledge of launch dates for an unrepresented launch system experiencing delays could provide the most up-to-date information.
- Where the spacecraft has been ordered, but the launch company has not been selected, the date the operator contracted for satellite readiness was used.
- Plans of existing satellite service operators were used as available.
- Plans of new or potential operators (i.e. growth in demand) were subject to the judgment of the individual subgroup members. It is this factor that led to the dispersions around the average forecast beginning in the year 1998.

Appendix 2: 1996 Near-Term COMSTAC Commercial GTO Mission Model

| | | 1995 | 1996 | 1997 | 1998 | TOTAL | Average Rate |
|------------|-------|---|--|---|--|------------|--------------|
| 101 | IAL = | 18 | 30 | 34 | 48 | 114 | 34.0 |
| | | • | 16 | 17 | • | - 42 | 14.0 |
| anespace | HEV | 1 Intelsat 706A | 1 Intelsof 707 A 3 Intelsof 709 | | Intetict 804 | | |
| | | | 1 Arabiot 2A | 1 Arabiot 28 | 1 Eutelsat-302 | | |
| | KV | 1 Brazil-Brazilsof B2 1 Eutelsaf-It F6-Hotbird 1 | 1 Argentino-Nahuel 3 | 1 Brazil-Brazilsot B3 | 1 Indio-Insat 25 | | |
| | | 1 Indio-Inset 2C | 1 Canada-TMI MSat M1 | | 1 Inmanat 305 1 Laoi-Estar 2 | | |
| | | 1 Japan-Nistar CS-A | 1 India-Insat 2D 1 Indianesia-Palapa C2 | | 1 US-World Space 1 | | |
| | 1 | 1 Luxembourg-SES Astro 1E 1 US-AT&T Teletor 402R | 1 Haly-Halsof 2 | | 1 US-World Space 2 | | |
| | 1 | 1 US-Hughes DBS 3 | 1 Japan-NStar CS-B | 1 Inmanat 304 | | | |
| | | 0 US-PAS 3R | 1 Thailand-Thaicom 3 | 1 Laos-LStor 1 1 Sweden-Sirius 2 | | | |
| | • | 1 US-PAS 4 | 1 Turkey-Turksat 1C 1 US-Echo Star 2 | 1 US-GE Americom GE2 | | | |
| | | | 1 US-PAS 3R | 1 US-PAS 7 | | | |
| | | | | 1 US-PAS DTH 6 | | | |
| | MLV | 0 Mataysia-MedSat 1 | 1 igrael-Armos 1 | | 1 Japan-BSat 1B | | |
| | , | *************************************** | 1 Malaysia-MeaSat 1 | 1 Japan-8Sat 1A | 1 UK-Skynet 4E | | |
| | | | 1 Malaysia-MeaSat 2 | | | - | |
| | | | | | | 13 | 43 |
| os . | | 5 | : . | 4 | 3 | i <u> </u> | |
| | HEV | 1 Intelect 704 | | | 1 Intelsat 805A 1 Intelsat 806A | | |
| | Ĺ | 1 Intelect 705 | | | | - | |
| | ILV | 1 AMSC-MSof M2 | 1 Eutoisat-Hollbird 2 | | 1 US-MCI/News Corp 2 | | |
| | ' | 1 Hughes-Galaxy 3R | 1 indonesia-Palapa C1 | 1 Japan-JSct 4 1 Japan-SCC-Superbird C | | | |
| | | 1 Japan-JSat 3 | 1 inmanat 301 1 inmanat 303 | 1 US-DBSC 1 | • | | |
| | | 1 | 1 US-GE Americom GE1 | | | | |
| | | | 1 US-TCI Tempo Sat 2 | | | _ | |
| | MLV | | | | | | |
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| ita | | 1 | 2 | 2 | 0 | 4 | 1.3 |
| - | HLV | | | | | | |
| | - | | 1 Kamasata | 1 Norway-Thor 2A | | | |
| | MEV | 1 Korecisch 1 0 Korecisch 2 | 1 KoreaSat 2 1 US-Hughes Galaxy 9 | 1 UK-Skynet 4D | | | |
| | | 0 100000 | | | | - | |
| 44 | | | 3 | 3 | 0 | . 6 | 2.0 |
| ing March | | | | 1 Chino-APStor 2R | | | |
| | MLV | 0 Intelsof 708A | 1 Inteliat 708A |] : CIEC/A-3/G-2# | | | |
| | ILV | 1 Chino-APStor 2 1 Chino-Asiasof 2 | | 1 Argentina-Nahuel 4 1 Philippine-Mabuhay 1 | | • | |
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| | | | 1 Chino-APStor 1A | | | | |
| | MLV | | 1 Chino-Chinosot 7 | | | | |
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| roton | | 0 | 3 | 5 | 0 | | 2.7 |
| | HLV | | | 1 Luxembourg-SES Astro 1G | | | |
| | | | | 1 US-AT&T Telstox 5 | | | |
| | ILV | | 1 Inmarsat 302 | 1 Chino-Asiasat 3 | | | |
| | | | 1 Luxembourg-SES Astra 1F | 1 US-MCI/News Corp.1 1 US-PAS 5 | | | |
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| | IIIV | <u></u> | | | | | |
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| | | 0 | | 5 | 36 | 41 | 13.7 |
| | | 0 | | 1 Luxembourg-SES Astro 1H | 1 Chino-APMT 1 | 41 | 13.7 |
| | ILV | | 0 | | 1 Chino-APMT 1 1 Indonesia-ACes 1 | 41 | 13.7 |
| | ILV | | | 1 Luxembourg-SES Astro 1H | 1 Chino-APMT 1 1 Indonesio-ACes 1 1 US-Hughes Galaxy 11 1 US-PAS 8 | 41 | 12.7 |
| | ILV | 0 | 0 | 1 Luxembourg-SES Astro 1H 1 Unnamed | 1 Chino-APMT 1 1 Indonesio-ACes 1 1 US-Hughes Galaxy 11 1 US-PAS 8 1 •Saudi Arabio-Safphone 1 | 41 | 13.7 |
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| | ILV | 0 | 0 | 1 Luxembourg-SES Astro 1H 1 Unnamed Not ordered Not ordered | 1 Chino-APMT 1 1 Indonesto-ACes 1 1 US-Hughes Galaxy 11 1 US-PAS 8 1 - Saudi Anabio-Satiphone 1 1 Lizembourg-SES Astro 2A 1 US-AT&T Telstar 6 | 41 | 13.7 |
| | ILV | 0 | 0 | Luxembourg-SES Astro 1H Unnamed Not ordered Not ordered Not ordered Not ordered | 1 Chino-APMT 1 1 Indonesio-ACes 1 1 US-Hughes Galaxy 11 1 US-PAS 8 1 - Saud Arabio-Scriphone 1 1 Luxembourg-SS Astro 2A 1 US-ATR T Testor 6 1 US-PAS 9 | 41 | 13.7 |
| | ILV | 0 | 0 | Lusernbourg-SES Astro 1H Unnormed Not ordered Not ordered Not ordered Not ordered 1 Eufelsot 3xx | 1 Chino-APMT 1 1 Indonesio-ACes 1 1 US-Hughes Galaxy 11 1 US-PAS 8 1 «Saudi Arabio-Safiphone 1 1 Lusembourg-SSS Astro 2A 1 US-ATS Teistor 6 1 US-PAS 9 1 Eutrebort-Horbird 5 | 41 | 13.7 |
| | HLV | 0 | | 1 Lusembourg-SES Astro 1H 1 Unnamed Not ordered Not ordered Not ordered 1 Eufebat 3ox 1 US-GE Americom GE3 | 1 Chino-APMT 1 1 Indonesio-ACes 1 1 US-Hughes Galaxy 11 1 US-PAS 8 1 - Saud Arabio-Sariphone 1 1 Luxembourg-SSS Astro 2A 1 US-ATST Testror 6 1 US-PAS 9 1 Eurebat-Hotbird 5 1 Singapore-ST 1 1 US-DSS 2 | 41 | 12.7 |
| | HLV | 0 | | Lusernbourg-SES Astro 1H Unnormed Not ordered Not ordered Not ordered Not ordered 1 Eufelsot 3xx | 1 Chino-APMT 1 1 Indonesto-ACes 1 1 US-Hughes Goldony 11 1 US-PAS 8 1 - Soud Arabio-Scriphone 1 1 Lusembourg-SES Astro 2A 1 US-AT&T Teletor 6 1 US-PAS 9 1 E-Meltor 16 1 US-PAS 9 1 E-Meltor 15 1 Singapore-ST 1 1 US-DESC 2 1 US-Maybres Goldony 10 | 41 | 13.7 |
| | HLV | • | | 1 Lusembourg-SES Astro 1H 1 Unnamed Not ordered Not ordered Not ordered 1 Eufelsat 3bx 1 US-GE American GE3 d 1 US-GE American GE4 | 1 Chino-APMT 1 1 Indonesio-ACes 1 1 US-Hughes Galany 11 1 US-PAS 9 1 «Souch Arabio-Scriphone 1 1 Lusembourg-SES Astro 2A 1 US-AT&T Teistor 6 1 US-PAS 9 1 Eurleuf-Hotbird 5 Singapore-ST 1 1 US-DESC 2 1 US-Hughes Galany 10 1 US-Orion 2 | 41 | 13.7 |
| | HLV | 0 | | 1 Lusernbourg-SES Astro IH 1 Unnormed Not ordered Not ordered Not ordered 1 Eutelsat 3ox 1 US-GE Americom GE3 d 1 US-GE Americom GE4 Not ordered | 1 Chino-APMT 1 2 Indonesto-ACes 1 1 US-Hugnes Goldon (1) 1 US-PAS 8 1 «Soud Arabio-Satiphone 1 1 Us-PAS 8 1 US-AT&T Teistor 6 1 US-PAS 9 1 Eurembourg-SSS Astro 2A 1 US-AT&T Teistor 6 1 US-PAS 9 1 US | 41 | 12.7 |
| | HLV | 0 | | 1 Lusernbourg-SES Astro 1H 1 Unnamed Not ordered Not ordered Not ordered 1 Eufeliat 3ox 1 US-GE Americom GE3 d 1 US-GE Americom GE4 Not ordered Not ordered | 1 Chino-APMT 1 1 Indonesio-ACes 1 1 US-Hughes Galany 11 1 US-PAS 9 1 «Souch Arabio-Scriphone 1 1 Lusembourg-SES Astro 2A 1 US-AT&T Teistor 6 1 US-PAS 9 1 Eurleuf-Hotbird 5 Singapore-ST 1 1 US-DESC 2 1 US-Hughes Galany 10 1 US-Orion 2 | A 1 | 13.7 |
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| | HLV | 0 | | 1 Lusernbourg-SES Astro 1H 1 Unnormed Not ordered Not ordered Not ordered 1 Eufebat 3xx 1 US-GE Americom GE3 d 1 US-GE Americom GE4 Not ordered Not ordered Not ordered Not ordered Not ordered Not ordered | 1 Chino-APMT 1 1 Indonesto-ACes 1 1 US-Hughes Golaxy 11 1 US-PAS 8 1 «Soud Arabio-Scriphone 1 1 UserPAS 8 1 «Soud Arabio-Scriphone 1 1 UserPAS 9 1 US-PAS 9 1 Canado-Telesort DTH 1 1 Canado-Telesort DTH 2 1 Chino-APS 10 3 1 Norway-Thor 28 | 4) | 13.7 |
| | HLV | 0 | | 1 Lusernbourg-SES Astro IH 1 Unnormed Not ordered Not ordered Not ordered 1 Eufestat 3ox 1 US-GE Americom GE3 d 1 US-GE Americom GE4 Not ordered | 1 Chino-APMT 1 1 Indonesto-ACes 1 1 Ish-Augnes Goldon / I1 IS-PAS 8 1 -Soud Arabio-Scriphone 1 I IS-PAS 8 1 -Soud Arabio-Scriphone 1 I IS-PAS 9 1 IS-AT&T Teletor 6 1 IS-PAS 9 1 Canado-Telesort DTH 1 1 Canado-Telesort DTH 2 1 Chino-APS 10 1 Chino-APS 10 1 Newson-Moreros 3/28 1 Newson-Moreros 3/28 1 Philippine-Machushy 2 | 4) | 12.7 |
| | HLV | 0 | | 1 Lusermbourg-SES Astro 1H 1 Unnamed Not ordered Not ordered Not ordered 1 Eufelsat 3ox 1 US-GE Americom GE3 d 1 US-GE Americom GE4 Not ordered | 1 Chino-APMT 1 1 Indonesto-ACes 1 1 US-Hughes Golaxy 11 1 US-PAS 8 1 «Soud Arabio-Scriphone 1 1 UserPAS 8 1 «Soud Arabio-Scriphone 1 1 UserPAS 9 1 US-PAS 9 1 Canado-Telesort DTH 1 1 Canado-Telesort DTH 2 1 Chino-APS 10 3 1 Norway-Thor 28 | 4) | 13.7 |
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| | HLV | 0 | | 1 Lusermbourg-SES Astro 1H 1 Unnamed Not ordered Not ordered Not ordered 1 LUS-GE Americom GE3 d 1 US-GE Americom GE4 Not ordered | 1 Chino-APMT 1 1 Indonesio-ACes 1 1 US-Hughes Golary 11 1 US-PAS 8 1 «Soud Anabio-Scriphone 1 1 UserPaS 8 1 «Soud Anabio-Scriphone 1 1 UserPaS 9 1 US-PAS 9 1 US-AT&T Teistor 6 1 US-PAS 9 1 US-PAS 9 1 US-DESC 2 1 US-Hughes Golary 10 1 US-Orton 2 1 US-Orton 2 1 Conado-Teiscort DTH 1 1 Conado-Teiscort DTH 2 1 Chino-APSico 3 1 Maxico-Morelos 3/2R 1 Monway-Thor 28 1 Philippine-Madoulinoy 2 1 US-G& Americom GES 3 3 Affittion-1996 Relatunches 1 *Archascri 2C | 4) | 13.7 |
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| | HLV | 0 | | 1 Lusermbourg-SES Astro 1H 1 Unnamed Not ordered Not ordered Not ordered Not ordered 1 LUS-GE Americom GE3 d 1 US-GE Americom GE4 Not ordered | 1 Chino-APMT 1 1 Indonesto-ACes 1 1 US-Hughes Golary 11 1 US-PAS 8 1 - Saud Arabio-Sarphone 1 1 Us-PAS 8 1 - Saud Arabio-Sarphone 1 1 Usenbourg-SES Astra 2A 1 US-AT&T Teistor 6 1 US-PAS 9 1 US-AT&T Teistor 6 1 US-PAS 9 1 US-AUGHES 1 US-Bughes Golary 10 1 US-Orion 2 1 US-Hughes Golary 10 1 US-Golfon 2 1 Canado-Teiesart DTH 1 1 Canado-Teiesart DTH 1 1 Canado-Teiesart DTH 2 1 Chino-APSior 3 1 Madisco-Mareiros 3/2R 1 Nonvay-Thor 28 1 Philippine-Madouhory 2 1 US-GE American GES 3 2 Affittion-1990 Relaunches 1 - Arabatar 2C 1 - Chino-Asiasart 4 3 - Egypt-Nessart 2 1 - Chino-Asiasart 4 3 - Egypt-Nessart 2 1 - Chino-Asiasart 4 3 - Egypt-Nessart 2 | 4) | 13.7 |
| | HLV | | | 1 Lusernbourg-SES Astro IH 1 Unnamed Not ordered Not ordered Not ordered 1 Eufebat 3ox 1 US-GE Americom GE3 d 1 US-GE Americom GE3 Not ordered | 1 Chino-APMT 1 1 Indonesto-ACes 1 1 US-Hughes Goldony 11 1 US-PAS 8 1 «Soud Arabia-Scriphone 1 1 Us-PAS 8 1 «Soud Arabia-Scriphone 1 1 Us-PAS 9 1 US-AT&T Teletor 6 1 US-PAS 9 1 US-AT&T Teletor 6 1 US-PAS 9 1 US-ACET 1 1 US-DESC 2 1 US-Mughes Goldony 10 1 US-Oston 2 1 US-Mughes Goldony 10 1 US-Oston 2 1 Condodo-Telesort DTH 1 1 Condodo-Telesort DTH 2 1 Philippin-Modologo C3 1 «Archard 2C 1 eIndonesta-Polapa C3 1 suppon-JCSort IP 3 1 suppon-JCSort | 4) | 13.7 |
| | HLV | • | | 1 Lusermbourg-SES Astro 1H 1 Unnamed Not ordered Not ordered Not ordered Not ordered 1 LUS-GE Americom GE3 d 1 US-GE Americom GE4 Not ordered | 1 Chino-APMT 1 1 Indonesto-ACes 1 1 Indonesto-ACes 1 1 US-Hughes Godary 11 1 US-PAS 8 1 - Saud Arabio-Sarphone 1 1 Lisembourg-SES Astra 2A 1 US-AT&T Teistor 6 1 US-PAS 9 1 US-AT&T Teistor 6 1 US-PAS 9 1 US-AE&T Teistor 1 1 US-DESC 2 1 US-Hughes Godary 10 1 US-Conado-Teissor DTH 1 1 Conado-Teissor DTH 2 1 Chino-APStor 3 1 Madsco-Marcios 3/2R 1 Norway-Thor 28 1 Philippine-Madoundry 2 1 US-GE American GES 3 2 Affittion-1999 Relationals 1 - Arabioar 2C 1 - Chino-Asiasan 4 3 - Egypt-Nessart 2 1 - Chino-Asiasan 4 3 - Egypt-Nessart 2 1 - Indonesto-Palapa C3 1 - Juppan-Superbird D 1 - Supeden-Palapa C3 1 - Juppan-Superbird D 1 - Supeden-Stud 3 | 4 | 12.7 |
| | HLV | • | | 1 Lutermbourg-SES Astro IH 1 Unnormed Not ordered Not ordered Not ordered Not ordered 1 Euteisat 3ox 1 US-GE Americom GE3 od 1 US-GE Americom GE4 Not ordered | 1 Chino-APMT 1 1 Indonesto-ACes 1 1 Ish-Augnes Goldon 11 1 US-HAS 8 1 -Soud Arabio-Satiphone 1 1 Us-PAS 8 1 -Soud Arabio-Satiphone 1 1 Us-PAS 9 1 Us-AT&T Teletor 6 1 Us-PAS 9 1 | 4) | 12.7 |
| IBD | HLV | O | | 1 Lutermbourg-SES Astro IH 1 Unnormed Not ordered Not ordered Not ordered Not ordered 1 Euteisat 3ox 1 US-GE Americom GE3 od 1 US-GE Americom GE4 Not ordered | 1 Chino-APMT 1 1 Indonesto-ACes 1 1 Indonesto-ACes 1 1 US-Hughes Godary 11 1 US-PAS 8 1 - Saud Arabio-Sarphone 1 1 Lisembourg-SES Astra 2A 1 US-AT&T Teistor 6 1 US-PAS 9 1 US-AT&T Teistor 6 1 US-PAS 9 1 US-AE&T Teistor 1 1 US-DESC 2 1 US-Hughes Godary 10 1 US-Conado-Teissor DTH 1 1 Conado-Teissor DTH 2 1 Chino-APStor 3 1 Madsco-Marcios 3/2R 1 Norway-Thor 28 1 Philippine-Madoundry 2 1 US-GE American GES 3 2 Affittion-1999 Relationals 1 - Arabioar 2C 1 - Chino-Asiasan 4 3 - Egypt-Nessart 2 1 - Chino-Asiasan 4 3 - Egypt-Nessart 2 1 - Indonesto-Palapa C3 1 - Juppan-Superbird D 1 - Supeden-Palapa C3 1 - Juppan-Superbird D 1 - Supeden-Stud 3 | 4) | 13.7 |

Not Included in COMSTAC Near-Term Commercial GTO Mission Model

| | 1995 | 1996 | 1997 | 1998 | TOTAL | Average Rate |
|------------------|---|--|--|---|----------|------------------------|
| 101 | AL = 23 | 26 | 40 | 30 | <u>%</u> | 32.0 |
| Ariane • | 1 ESA-ERS 2 1 ESA-60 1 France-Helios 1 11 France-Telecom 2C | : ESA-European Cluster 1 ESA-Recovery Module 1 France-Telecom 2D | 1 France-Spot 4 1 Eumenat-Meteosat(MOP | 1 ESA-Arierns 1 4) 1 France-Helios 18 | - | |
| Attas | 1 ESA-SOHO 1 US-AF-OSCS 3-05 1 US-N-UHF-FEHF FOA 1 US-N-UHF-FEHF FOS 3 US-N-UHF-FEHF FOO 1 US-N-UHF-FOO | ? ESA-SAX-Astronomy 1 US-A-UMF/EHF F07 | I US-AF-COII UP MLV-7 I US-AF-DSCS 3-06 I US-NASA/NOAA-Goes K | 1 US-N-UHF/EHF FOB 1 US-N-UHF/EHF FOP 1 US-N-UHF/EHF F10 1 US-NASA-EOS AM | _ | |
| Delta | 1 Canada-Radarsat | 1 Iridhum 01 - 5 | I GlobatStar 1 - 04 | 1 US-AF-GPS 2R-04 | | |
| | 1 US-NASA-XTE | 1 Indiam 02 - 5 1 US-AF-GPS 2-Block 2-07 1 US-AF-GPS 2-Block 2-10 1 US-AF-GPS 2-Block 2-10 1 US-AF-Middourie Space Exp 1 US-NASA-Middourie Space Exp 1 US-NASA-MIGGOURIE Pathfinder 1 US-NASA-NEAR 1 US-NASA-NEAR 1 US-NASA-Polar | Globalstar 2 - 04 Indulum 03 - 5 Indulum 04 - 5 Indulum 04 - 5 Indulum 05 - 5 Indulum 05 - 5 Indulum 07 - 5 Indulum 08 - 5 IUS-AF-GPS 2-Block 2-08 US-AF-GPS 2-Block 2-09 US-AF-GPS 27-02 US-A | 1 US-AF-GPS 24-05. 1 US-AF-GPS 24-06 1 US-AF-GPS 24-06 1 US-AF-GPS 24-07 1 US-AV-SA-Landsof 7 1 Infaltem 09 - 03 1 Infaltem 10 - 03 1 US-AF-SMTS/Britischt Eyes | | |
| Japan | 1 Japan-SMS 1 Japan-SFU | 1 Japan-ADEOS | 1 Japan-Commets 1 Japan-TRMM 1 Japan-ETS-7 | | | |
| Long March | | TjChino-DFH 302 1 Chino-Fengyun 2A | China-DFH 303 1 China-Onentstar 1 China-Sinosar 1 | 1 Brazi-CBERS 1 11 Chino-DFH 401 11 Chino-DFH 402 11 Chino-Snosot 2 1 GlobalStar 01 - 12 | - | • |
| Proton | , | | | | - | |
| | ∏Russio-GALS 2 | 1: Russio-Express 02 1: Russio-Gorizont 31 1: Russio-Gorizont 31R 1: Russio-Gorizont 32 | I Indiam 01 - 7 I Indiam 02 - 7 I Indiam 03 - 7 I Russio-Express 03 I Russio-Express 04 I Russio-Yomal I Russio-Yomal | 1 Russic-Express 05 1 Russic-Express 06 1 Russic-Express 07 1 Russic-GALS 3 | | |
| Zenit | 7 Domestic Requirements | 3 Domestic Requirements | 5 Domestic Requirements | 6 Domestic Requirements | - | |
| | | |) GlobalStar 01 - 12) GlobalStar 02 - 12) GlobalStar 03 - 12 | | | |
| тво | Domestic Requirements | 1 Domestic Requirements | Domestic Requirements | 7 Domestic Requirements | - | |
| Note: Legend: | | e spacecraft as single requirements or spacecraft as single requirements or spacecraft as spacecraft | | iilure | | |
| | Name Spacecraft not include | f MEO Launch Mission - Usually involved ded in all members models commercial communication servic | | em manufacturer, captive laur | nch | |
| | 1995 | 1996 | 1997 | 1998 | TOTAL | 3 Year Average Rate |
| TOTAL SPAC | | 56 | 76 | 78 . | 210 | 70.0 |
| | clusters counted as one | | 57 | 15 | 80 | 26.7 |
| Long March 2C | Small | | 1 Iridium 01 - 2 1 Iridium 02 - 2 1 Iridium 03 - 2 | 1 Iridium 04 - 2 1 Iridium 05 - 2 1 Iridium 06 - 2 | - | |

Report of the COMSTAC Technology & Innovation Working Group

COMMERCIAL SPACECRAFT MISSION MODEL UPDATE

May 1996

Appendix 3 1988 - 1995 Mission Model - History

Commercial Space Transportation Advisory Committee (COMSTAC)
Office Of Commercial Space Transportation
Federal Aviation Administration
U.S. Department of Transportation

Appendix 3:COMSTAC Mission Model - History Commercial GTO Mission Model

| | | 1988 | 1989 | 1990 | 1661 | 1992 | 1993 | 1994 | 1995 | TOTAL | Average Rate |
|-------------|----------|---|--|--|--|--|--|--|---|---------------|-----------------|
| | TOTAL - | 6 | 8 | 81 | * | 11 | 01 | 18 | 81 | 112 | 14.0 |
| Arlanespace | | 6 | , | ٥ | 0 | • | 9 | 12 | 6 | 1/ | 6.8 |
| | Ж | | Intelsal 602 Japan-JSS JCSal 1 Japan-SCC Superbird A | Japon-SCC Superbird B US-Comsat SBS 6 | Conado-Telesat Anik El Conado-Telesat Anik E2 Inlesat 601 Inlesat 605 Uxemboura SES Astra 18 | US Hughes Goloxy 7 Japon-SCC Superbild Japon-SCC Superbild | US Hughes Golaxy 7 Intelect 701 Japan-SCC Superbird B Luxembourg-SES Astra 1C Japan-SCC Superbird A Mexico-Solidaridad 1 US Hughes DBS 1 US Hardhes Golaxy 4 | Intelsat 702 | Intelsot 706A Japan-NSlar CS-4A | | |
| | 1[4 | Intekat 513A Germany-DB Luxambourg-SES Astra 1 Intekat 515A | Germany-DBP TVSat 2 Intelsat 515A | Eutelsat 201 | Eutelsot 202 Italy-Italsot 1 | Eutelsof 204 India-Insal 2A | India-Insat 28 Social-Hispasot 18 | Brazil-Brazilsot B1 | Brazil-Brazilsal 82 | , | |
| | | D | Sweden-SSC Tele X | | | Spain-Hispasat 1A | a peode | Luxembourg-SES Astra 1D India-Insol 2C Mexico-Solidaridad 2 Luxembourg-SES | ID India-Insol 2C Luxemboura-SES Astro 1E | | |
| | | | | | | | | US Panamsot 3 | US-Af&T Telstor 402R | | |
| | | | | | | | | Turkey-Turksot 1B | US-PAS 4 | | |
| | | | | | | | | US-Panamsat 2 | 7 | | |
| | ≥ ¥ | | Germany-DBP DFS 1 | Germany-DBP DFS 2 Japan-Nasda BS 2X | Inmarsat 2 F3 | US-GE Salcom C3 Arabsal 1C | Thailand-Thalcom 1 | Thailand-Thalcom 2 Jopan-NHK BS 3N | | | |
| | | UK-Skynel 48 US-Comsol 58S.5 | - | UK-Skynet 4C US-GE Satcom C1 | | Inmarsal 2 F4 | | | | | |
| | | US-GIE GStor 3 US-GIE Spocenet 3R US-Ponomsat 1 | , | US-GIE GStar 4 US-Hughes Galoxy 6 | | | | | | | |
| | | | | | | | | | | 7 | |
| Attas | | 0 | 0 | 0 | 2 | 9 | - | 9 | ş | 14 | 9.1 |
| | ¥ | | | | | | | Intelsat 703 | Intelsal 704 | | |
| | 2 | | | | Eutelsor 203 | Intelsat K1 | US-AT&T Telstar 401 | US-Hughes DBS 2 | US-Hughes Goloxy 3R | 1 | |
| | | | | | | | | US-Orion 1 | Japan-JSat 3 US-AMSC MSat M2 | | |
| | NIV. | | | | Japan-NHK BS 3H | US Hughes Golaxy 1R US Hughes Golaxy 5 | | | | ī | |
| | | | | | | | | | | 1 | |
| Della | | 0 | - | 4 | | 6 | - | - | 1 | 15 | 1.9 |
| | À. | | UK-BSB/Marcopola 1 | India-Insat 10 Indonesia-Palapa 803 | Inmarsat 2 F2 NATO 4A | Germany-DBP DFS3 Indonesia-Pologo Bd | NAIO 4B | US-Hughes Galaxy IR/2 | KoreaSat 1 | | |
| |] | | | Inmaisat 2 F 1 UK-BSB/Marcopolo 2 | US-GE Satcom(Aurora) C5 US-GTE Spacenet (ACS 2) 4 | | | | | | |
| | | | | | | | | | | | |
| Titan 3 | | 0 | 0 | 7 | 0 | 0 | 0 | 0 | 0 | 4 | 9.6 |
| | <u>}</u> | | | Intelsat 603 Intelsat 604 | | | | | | | |
| | ≧ | | | Japan-JCSat 2 | | | | | | 1 | |
| • | MIN | | | UK-Skynel 4A | | | | | | \Box | |
| dozel Merch | | • | | - | | | | | | | |
| Tour Marie | HI. | • | 0 | _ | 0 | 2 | 0 | 2 | 6 | | 6. |
| | <u> </u> | | | , | | Australia-Optus B1 Australia-Optus B2 | | Australia-Optus 83 | China-APStar 2 Chira-Asiasat 2 | T | |
| | ¥ | | | China-Aslasof 1 | | | | China-APStar 1 | י ייייי פייייי | $\overline{}$ | 8/14/96 |

Appendix 3:COMSTAC Mission Model - History

Not included in Commercial GTO Mission Model - Utilized Commercial Launch Service Vehicles

| | 1988 | 1989 | 1990 | 1661 | 1992 | 1993 | 1994 | 1995 | TOTAL | Rate |
|-----------------------------|---|---|--|--|--|--|---|--|-------|------|
| TOTAL - | - | 13 | 92 | 40 | 14 | 14 | 10 | 15 | 100 | = |
| | ESA-Meteosot 3 France-ToF 1 France-Telecom 1C | IESA-Oympus I IESA-Hipporcos ESA-Meteosal 4 | France Spot 2 France 10F 2 | ESA-ErS 1 ESA-Meleosot 5 Fronce-Telecom 2A | France-Telecom 28 NASA-TOPEX | Eumelsal-Meleosal 6 France-Spot 3 | | ESA-ERS 2 ESA-ISO France-Heltos 1 France-Telecom 2C | | |
| • | US-AF DAKP FO9 US-NOAA 11 | US Navy Fikalcom 8 | US AF DMSP F10 US AF Stocked US NASA/AF CRESS | NOAA 12 US-AF DMSP F11 | USAF-DSCS 3 B01 USAF-DSCS 3 B02 | NOA 13 US-AF DSCS 3-03 US-AF DSCS 3-04 USA-UHF F0T | US-AF DMSP F12 US-NOW UHF F03 US-NOAA 14 US-NOAA Goes 8 | ESA-SOHO US AF DSCS 3-05 US NASA/NOA GOBS J US NOAV UHFEHF FOB US NOAV UHFFEHF FOB US NOAV UHFFEHF FOB | | |
| | US-AF DMA3-thusive | US-AF DMA3-IhrustVecExt US-AF Cos Bligard Exp US-AF Della Star US-AF GPS Navstar 01 US-AF GPS Navstar 03 US-AF GPS Navstar 03 US-AF GPS Navstar 04 US-AF GPS Navstar 04 | Germany-Rosal-X Ray US-AF GPS Navetor 06 US-AF GPS Navetor 07 US-AF GPS Navetor 09 US-AF GPS Navetor 10 US-AF GPS Navetor 10 US-AF Relay-Mitrockxp | US-AF GPS-Mavstor 11 | Jopan-Geotal US-AF GPS Navatar 12 US-AF GPS Navatar 13 US-AF GPS Navatar 14 US-AF GPS Navatar 15 US-AF GPS Navatar 15 US-AF GPS Navatar 15 US-AF GPS Navatar 16 US-AF GPS Navatar 16 US-AF GPS Navatar 16 US-AF GPS Navatar 16 | USAF GPS 2 BIK 2 0 1 USAF GPS 2 BIK 2 0 2 USAF GPS 2 BIK 2 0 3 USAF GPS 2 BIK 2 0 3 USAF GPS 2 BIK 2 0 5 USAF GPS 2 BIK 2 0 5 | NASA-WING US-AF GPS 2 Block 2 06 | Conodo-Radarsat US-NASA XIE | | |
| | US-AF 1llan 2 | US-AF Illan 2 | | | US-AF Titon 2 | US-NASA Londsof & | US-NASA Clementine | | | |
| | Japan-CS 3A Japan-CS 3B | Japan-GMS 4 | Japan-BS 3A Japan-MOS 1B | Japan-85 38 | Japan-JERS | | Japan-EIS 6 | Japon-GMS Japon-SFU | | |
| Proton - (Western Use Only) | (Aju | | | | | | Russia-Rimsol Express | Russia-GALS 2 | | |
| Long March | Chino-DDH 201 Chino-DFH 202 | | Ching-DFH 203 | Chino-DFH 204 | П | | China-DFH 301 | | | |

Appendix 3:COMSTAC Mission Model - History

Not Included in Commercial GTO Mission Model - Did Not Utilized Commercial Launch Service Vehicles

| | 1988 | 1989 | 1990 | 1661 | 1992 | 1993 | 1994 | 1995 | TOTAL * | % Average Rate | age 1e |
|----------------------------|------------------------------------|--|--|---|---|----------------------------------|--|--|---------|-------------------|-----------|
| TOTAL - | | 6 | 9 | 7 | , | 0 | * | 9 | 48 | 0.9 | |
| Shuffle | US NASA TDRS C | US-DoD US-DoD US-NASA GOREO US-NASA MOGERON US-NASA TDRS D | US-DOD US-DOD US-DOD US-NASA ASIRO 1 US-NASA HADDIB US-NASA HADDIB US-NASA UN-SSB8S US-NOW SYNCOM IV-5 | US-POD US-NASA GRO US-NASA IDISE US-NASA UARS US-SDIO | ESA-Eureko Intekst 603 Reboost US-Dod US NASA Logeos II US NASA ESS | US NASA TORS F US NASA TORS F | | US NASA IDIRSF US-NASA WSF 2 | | | · |
| Thon III & IV | US-AF Titon 34D US-AF Titon 34D | US-AF DSCSII-16 US-AF DSCSII/4 US-AF DSP 14 US-AF Ilton 34D | US-AF IIfon 4 | US-AF liton 4 US-AF liton 4 | US-AF Itlan 4 US-NASA Mars Observer | US-AF liton 4 | US-AF DSP 17 US-AF Milstor DFS 1 US-AF 1lton 4 | US-AF Mistor DFS 2 US-AF Titon 4 US-AF Titon 4 US-DoD | | | |
| Legend: | Spacecraft fallec | Spacecraft falled to reach operaling status as planned | alus as planned ng operaling status | | | | | | | | |
| TOTAL SPACECRAFF LAUNCHED= | FF 24 | 30 | 17 | 53 | 38 | 27 | 32 | 39 | 560 | 32.5 | 3 |
| FAILURES | 8 | 8 | 6 | 7 | e | 6 | so. | - | 2 | 2.6 | • |
| | 8.3% | 6.7% | 7.3% | %6'9 | 7.9% | 11.1% | 15.6% | 2.6% | 8.1% | 8.1% | <u>*</u> |

Statistical Summary of Addressable GEO Spacecraft MANUFACTURERS

| TOTAL LAUNCHED= | ٥ | · 6 0 | 81 | 7 | 17 | 2 | 18 | 10 | | | 14.0 |
|-----------------|---|-----------------|----|----|----|----|-----|----|----|-----|------|
| LAUNCHER | | | | | | | | | | | |
| S | 0 | - | €0 | • | • | 2 | 4 | | | | 1.4 |
| EUROPEAN - | ۰ | , | ۰ | 80 | ۰ | 60 | 12 | • | | | 8.9 |
| OTHER - | 0 | 0 | _ | 0 | 2 | 0 | 7 | | | | 0.1 |
| SPACECRAFI | | | | | | | | | | | |
| ı sn | • | • | 12 | • | = | €0 | 14 | 92 | | 70% | 9.8 |
| EUROPEAN = | ~ | 8 | LO | • | s. | 2 | . 4 | _ | 28 | | 3.5 |
| OTHER - | _ | 0 | _ | 7 | - | _ | 0 | _ | | | 6.0 |
| | | | | | | | | | | | |

OPERATQRS